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Geodesics in Schwarzschild, Reminders:

$$\left(\frac{dr}{ds}\right)^2 = -\left(\lambda + \frac{J^2}{r^2}\right)\left(1 - \frac{2m}{r}\right) + E^2. \quad (1)$$

where $\lambda = 0$ for null geodesics, and $\lambda = 1$ for timelike ones, and

$$\frac{dt}{ds} = \frac{E}{1 - \frac{2m}{r}}. \quad (2)$$

$$\frac{d\varphi}{ds} = \frac{J}{r^2}. \quad (3)$$

$u := m/r$, if $J \neq 0$:

$$\left(\frac{du}{d\varphi}\right)^2 = \frac{m^2 E^2}{J^2} - u^2(1 - 2u) - \frac{\lambda m^2(1 - 2u)}{J^2}, \quad (4)$$

$$\frac{d^2 u}{d\varphi^2} = -u + 3u^2 + \frac{\lambda m^2}{J^2}. \quad (5)$$

§II.9 Massive test particles: advance of the perihelion/periastron

Periastron: the point at which the orbit is closest to the star.

When the star is our sun: *perihelion*.

Aim of this section: compare the motion in the Schwarzschild metric with the Keplerian orbits, at distances large compared to m .

The orbit of the earth ($r = r_{\oplus}$, $J = J_{\oplus}$) around the sun (so that $m = M_{\odot}$) we have

$$\frac{2M_{\odot}}{r_{\oplus}} = 2u \sim 10^{-8}, \quad \frac{J_{\oplus}^2}{r_{\oplus}^2} = J_{\oplus}^2 M_{\odot}^{-2} u^2 \sim 10^{-8}, \quad \frac{M_{\odot}}{J_{\oplus}^2} \sim 10^{-8}.$$

(m is the mass of the central body, J is the angular momentum per unit mass of the orbiting body)

Exercise: check the following numbers, work out the corresponding numbers for Mars and Venus

Comparison with Newton, once again:

Return to the Schwarzschildian (1) with $\lambda = 1$

$$\begin{aligned} \left(\frac{dr}{ds}\right)^2 &= -\left(1 + \frac{J^2}{r^2}\right) \left(1 - \frac{2m}{r}\right) + E^2 \\ &= \underbrace{E^2 - 1 - \frac{J^2}{r^2} + \frac{2m}{r}}_{\text{leading contribution}} + \underbrace{\frac{2mJ^2}{r^3}}_{\frac{m}{r} \times \frac{J^2}{r^2}}. \end{aligned} \quad (6)$$

Newtonian test body with unit mass, and with angular momentum J_N , conservation of angular-momentum leads to the following formula for the Newtonian energy E_N

$$E_N = \frac{1}{2}(\dot{r}^2 + r^2\dot{\varphi}^2) - \frac{m}{r} = \frac{1}{2}\left(\dot{r}^2 + \frac{J_N^2}{r^2}\right) - \frac{m}{r},$$

hence

$$\left(\frac{dr}{dt}\right)^2 = 2E_N - \frac{J_N^2}{r^2} + \frac{2m}{r}. \quad (7)$$

Identical equations

1. if $s = t$ and if
2. we neglect the $\frac{2mJ^2}{r^3}$ term in (6),
3. identify $E^2 - 1$ with $2E_N$
4. identify J with J_N (recall unit mass)

Similarly the u equation (4) with $\lambda = 1$,

$$\left(\frac{du}{d\varphi}\right)^2 = \frac{m^2 E^2}{J^2} - u^2(1 - 2u) - \frac{m^2(1 - 2u)}{J^2}, \quad (8)$$

should be compared with its version where the u^3 terms have been dropped out:

$$\begin{aligned} \left(\frac{du}{d\varphi}\right)^2 &= \frac{m^2 E^2}{J^2} - u^2 - \frac{m^2(1 - 2u)}{J^2} \\ &= \frac{m^2}{J^2}(E^2 - 1 + 2u) - u^2 \end{aligned}$$

Recall the Newtonian equations from a previous lecture, where now m_0 the mass of the particle and G should be set to one:

$$\begin{aligned} \left(\frac{du}{d\varphi}\right)^2 &= \frac{2m_0 m^2 E_N}{J_N^2} - u^2 + \frac{2Gm_0^2 m^2}{J_N^2} u \\ &= \frac{2m^2 E_N}{J_N^2} - u^2 + \frac{2m^2}{J_N^2} u \\ &= \frac{m^2}{J_N^2}(2E_N + 2u) - u^2. \end{aligned} \quad (9)$$

and (Newton):

$$\frac{d^2 u}{d\varphi^2} = -u + \frac{Gm_0^2 m^2}{J_N^2}$$

$$= -u + \frac{m^2}{J_N^2} . \quad (10)$$

The u equation for Schwarzschild (5) with $\lambda = 1$ again:

$$\frac{d^2 u}{d\varphi^2} = -u + 3u^2 + \frac{m^2}{J^2} . \quad (5)$$

The Newtonian solution derived previously:

$$u_0 = \frac{m^2}{J^2} (1 + e \cos \varphi) , \quad (11)$$

satisfies $\dot{u} = 0$ at $\varphi = 0$, and u has a *maximum* at $\varphi = 0$.

In (5) for u with $\lambda = 1$ we insert $u = u_0 + v$, where both u_0 and v are small, approximate

$$u^2 \approx u_0^2$$

to get

$$\frac{d^2 v}{d\varphi^2} + v \approx 3 \frac{m^4}{J^4} (1 + e \cos \varphi)^2 . \quad (12)$$

This can be integrated with e.g. MATHEMATICA to obtain

$$v(\varphi) \approx \frac{m^4}{J^4} \left[-(3 + e^2) \cos \varphi + 3 \left(1 + \frac{e^2}{2} \right) - \frac{e^2}{2} \cos 2\varphi + 3e \varphi \sin \varphi \right] ,$$

where the free integration constants have been chosen so that $v(0) = v'(0) = 0$. Thus,

$$u(\varphi) \approx \underbrace{\frac{m^2}{J^2}(1 + e \cos \varphi)}_{u_0} + \frac{m^4}{J^4} \left[-(3 + e^2) \cos \varphi + 3 \left(1 + \frac{e^2}{2} \right) - \frac{e^2}{2} \cos 2\varphi + 3e\varphi \sin \varphi \right] .$$

We seek the *perihelion*, the point of closest approach to the center, hence a maximum of u . Now

$$\begin{aligned} \partial_\varphi u(\varphi) \approx & -\frac{m^2}{J^2} e \sin \varphi \\ & + \frac{m^4}{J^4} \left[(3 + e^2) \sin \varphi + 2\frac{e^2}{2} \sin 2\varphi \right. \\ & \left. + 3e \sin \varphi + 3e\varphi \cos \varphi \right] . \end{aligned}$$

so this is indeed an extremum $\varphi = 0$

$$\partial_\varphi u = 0 ,$$

and it is clear from the Newtonian expression (11) that this is a maximum of u for m/J small enough. The next maximum will be at $\varphi = 2\pi + \gamma$, with γ small:

$$\begin{aligned} 0 &= \partial_\varphi u \\ &\approx -\frac{m^2}{J^2} e \sin \varphi + \frac{m^4}{J^4} \underbrace{\left[(3 + e^2) \sin \varphi + e^2 \sin 2\varphi + 3e \sin \varphi \right]}_{\approx \gamma \ll 1} \\ &\quad + 3e \underbrace{\varphi}_{\approx 2\pi} \underbrace{\cos \varphi}_{\approx 1} \\ &\approx -\frac{m^2}{J^2} e \gamma + 6 \frac{m^4 e \pi}{J^4} , \quad \text{since } \sin \varphi = \sin(2\pi + \gamma) \approx \gamma . \end{aligned}$$

We have thus obtained, in SI units, if $e \neq 0$,

$$\gamma \approx 6 \frac{m^2 \pi G}{J^2 c^2} .$$

This is the *perihelion advance* predicted by general relativity. For Mercury this is sometimes called the *perihermion* advance, and equals about (exercice)

$$40'' \text{ per century} .$$

A more detailed calculation gives the observed value of $43''$ per century. This value was a puzzle to astronomers at the beginning of the twentieth century.

For the Taylor-Hulse pulsar the advance is around 4° per year, which has to be corrected to account for the gravitational waves

§II.10: Geodetic precession

1. In Newtonian theory, consider a gyroscope rotating along a given direction, without any forces acting on it. Then the direction of rotation, and its velocity will remain unchanged. We can thus associate to a gyroscope a vector \vec{s} such that

1. \vec{s} points in the direction of rotation, and
2. $|\vec{s}|$ describes the speed of rotation.

Then we have

$$\frac{d\vec{s}}{dt} = 0 .$$

2. Let a gyroscope travel on a geodesic $\gamma(\tau)$ in Minkowski space-time, without any forces acting on it. In a Newtonian inertial frame the direction of rotation of the gyroscope will not change. It should then also be true that in a special relativistic inertial frame the direction of the gyroscope will not change either.

In inertial coordinates adapted to γ ,

$$\gamma(\tau) = (\tau, \vec{0}) ,$$

the four-vector $s = (0, \vec{s})$ satisfies then, along γ , the set of equations

$$\frac{Ds}{d\tau} = 0 , \quad \eta(u, s) = 0 , \quad (13)$$

with η the Minkowski metric.

This is related to the notion of **non-rotating frames**.

3. One can invoke the correspondence principle to argue that

$$\frac{Ds}{d\tau} = 0, \quad g(u, s) = 0, \quad (14)$$

is the right general-relativistic equation to describe gyroscopes or, equivalently, **non-rotating local frames**.

Remember that

$$\frac{Ds^\mu}{d\tau} := \frac{ds^\mu}{d\tau} + \Gamma_{\alpha\beta}^\mu s^\alpha \dot{\gamma}^\beta.$$

§II.10.1: The parallel transport equation

Consider a geodesic $\gamma(\tau)$ with tangent vector $u \equiv \dot{\gamma} := d\gamma/d\tau$. As argued above, a gyroscope along γ can be described by a vector s orthogonal to u which is *parallel transported along γ* :

$$g(s, u) = 0, \quad \frac{Ds}{d\tau} = 0. \quad (15)$$

General facts about the **parallel transport equation**:

$$\frac{DX}{d\tau} = 0$$

Let X and Y be two vectors which are parallelly propagated along *any curve* $\gamma(\tau)$, not necessarily geodesic. Thus

$$\frac{DX}{d\tau} = 0 = \frac{DY}{d\tau}, \quad \text{with} \quad \frac{DX^\alpha}{d\tau} := \frac{dX^\alpha}{d\tau} + \Gamma_{\mu\nu}^\alpha X^\mu \frac{d\gamma^\nu}{d\tau}. \quad (16)$$

Now, for any vectors, regardless of parallel transport,

$$\frac{d}{d\tau}(g(X, Y)) = \frac{d}{d\tau}(g_{\mu\nu} X^\mu Y^\nu)$$

$$= \underbrace{\frac{dg_{\mu\nu}}{d\tau}}_{g_{\mu\nu,\sigma}\dot{\gamma}^\sigma} X^\mu Y^\nu + g_{\mu\nu} \frac{dX^\mu}{d\tau} Y^\nu + g_{\mu\nu} X^\mu \frac{dY^\nu}{d\tau} .$$

Since

$$0 = \nabla_\sigma g_{\mu\nu} = \partial_\sigma g_{\mu\nu} - \Gamma_{\mu\sigma}^\lambda g_{\lambda\nu} - \Gamma_{\nu\sigma}^\lambda g_{\lambda\mu} ,$$

equivalently,

$$\partial_\sigma g_{\mu\nu} = \Gamma_{\mu\sigma}^\lambda g_{\lambda\nu} + \Gamma_{\nu\sigma}^\lambda g_{\lambda\mu} ,$$

we obtain

$$\begin{aligned} \frac{d}{d\tau}(g(X, Y)) &= \underbrace{\frac{dg_{\mu\nu}}{d\tau}}_{(\Gamma_{\mu\sigma}^\lambda g_{\lambda\nu} + \Gamma_{\nu\sigma}^\lambda g_{\lambda\mu})\dot{\gamma}^\sigma} X^\mu Y^\nu + g_{\mu\nu} \frac{dX^\mu}{d\tau} Y^\nu + g_{\mu\nu} X^\mu \frac{dY^\nu}{d\tau} \\ &= g\left(\frac{DX}{d\tau}, Y\right) + g\left(X, \frac{DY}{d\tau}\right) . \end{aligned}$$

Summarising:

$$\boxed{\frac{d}{d\tau}(g(X, Y)) = g\left(\frac{DX}{d\tau}, Y\right) + g\left(X, \frac{DY}{d\tau}\right) .}$$

For parallel-transported vectors,

$$\boxed{\frac{d}{d\tau}(g(X, Y)) = 0 .}$$

Thus, the angle between the vectors is preserved along the curve.

In particular if $X \perp Y$ at one point, then it will be so along the whole curve.

Similarly if $g(X, X) = \pm 1$ at one point, it will be so along the whole curve.

Paralelly transporting an ON basis e_a of the tangent space $a = 0, \dots, n$ at some point of γ we will obtain an ON basis of $T\mathcal{M}$ at any point lying on the image of γ .

If γ is an affinely parameterised geodesic, then the tangent vector u is parallel along γ . So, for a gyroscope, the first condition

$$g(s, u) = 0$$

in (15) is consistent with the evolution equations for s and u . Equivalently, if $g(s, u)$ vanishes at one point of the geodesic, it will vanish everywhere.